# Fabrication challenges and solutions a. k. a. Design and manufacture of mirrors, and active optics

Buddy Martin
Steward Observatory Mirror Lab

#### Outline

- What makes a good mirror?
- Modern mirror concepts
  - thin solid mirrors
  - segmented mirrors
  - lightweight mirrors
- Honeycomb mirrors
  - design
  - casting
- Optical manufacture
  - requirements
  - aside on active optics and model fitting
  - fabrication (2/17/11 lecture at Mirror Lab)
    - machining
    - polishing
  - measurement (2/17/11 lecture at Mirror Lab)
    - interferometry
    - null correctors
    - GMT measurements

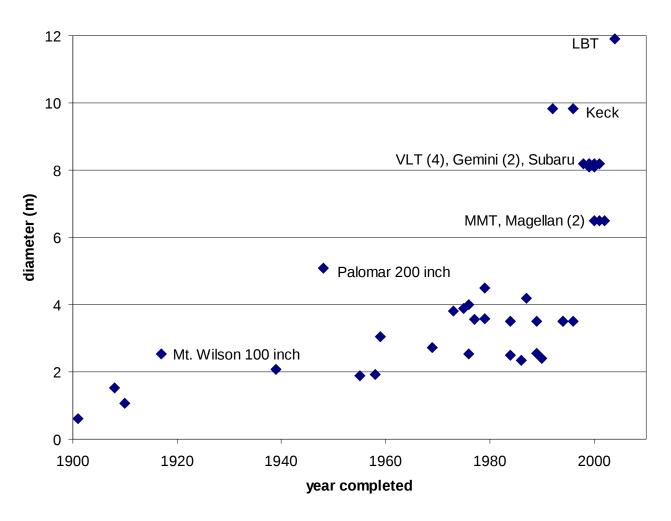
### What makes a good mirror?

- Fundamental requirement is to deliver a "good" wavefront to focal plane in almost all conditions.
  - Hold its shape to a fraction of a wavelength
  - Be smooth to a *small* fraction of a wavelength on small scales
  - Contribute little to local seeing (temperature gradients in air)
- Stiffness against wind: bending stiffness prop. to E t<sup>3</sup>
  - E = Young's modulus, t = thickness
  - more complicated for lightweight mirror
  - Stiffness of honeycomb mirrors is about right; use this as baseline for comparisons.
- Stiffness against gravity: bending stiffness prop. to E  $t^2/\rho$ 
  - Again, use honeycomb mirrors as baseline.
- Thermal distortion: displacement =  $\alpha \Delta T$  t for "swelling"

curvature = 
$$\alpha \Delta T / t$$
 for bending

- $\alpha$  = thermal expansion coefficient,  $\Delta T$  = temperature variation within mirror
- "Mirror seeing" prop. to T  $T_{air} \sim dT_{air}/dt \cdot \tau$ 
  - $dT_{ii}/dt$  = rate of change of air temperature
  - $\tau$  = mirror's thermal time constant ~ c ρ  $t^2 / k$ 
    - c = specific heat, k = thermal conductivity, t = thickness
  - Becomes a problem for T  $T_{air} > \sim 0.3$  K,  $\tau > \sim 1$  hr
  - For glass or glass-ceramics, want t < 5 cm</li>
- Bottom line: Mirror should be stiff & light, have low thermal expansion & short thermal time constant.

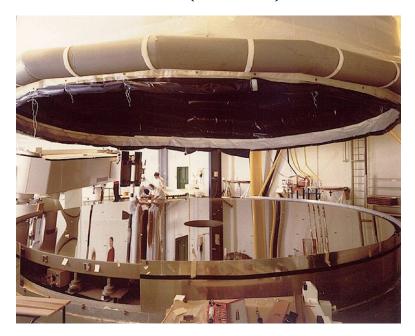
## Optical telescopes

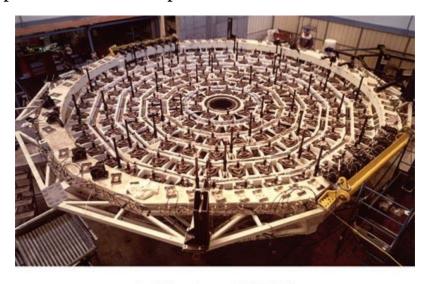


- Hale Telescope at Palomar used first large lightweighted mirror.
- Most powerful telescope for 45 years because of difficulty making a larger mirror that would not distort due to its weight and thermal inertia.

#### New mirror concepts after 1980

- 3 solutions emerged ~1980:
- Thin, solid mirrors whose shape is controlled by active optics
  - Active optics concept by Ray Wilson and colleagues in Europe
  - Concept:
    - · Replace stiffness by active control of shape
    - Reduces mass and thermal inertia (somewhat) with 175 mm thick mirror
  - Technology:
    - Zerodur glass ceramic and ULE glass, both with near zero expansion coefficient
    - Precise active mirror supports
    - Wavefront sensors similar to those used for adaptive optics
  - ESO VLT (4 x 8.2 m), 2 Gemini telescopes, Subaru telescope

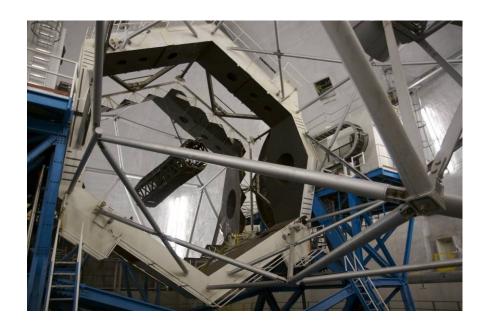


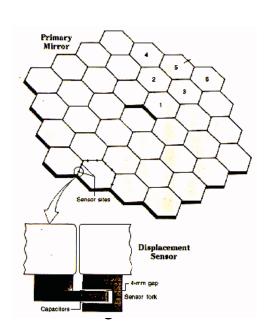




## Segmented mirrors

- Developed by Jerry Nelson and colleagues at UC
- Concept:
  - Achieve continuous optical surface by active control of position of small segments.
  - Reduces mass and thermal inertia even more than thin solid mirror (75 mm vs 175 mm)
- Technology
  - Precise segment positioning actuators
  - Precise segment-segment displacement sensors (capacitive)
  - Occasional wavefront measurement of segment phasing
- Used for Keck, Hobby-Eberly, Grantecan, SALT
- To be used for TMT (30 m), ESO ELT (42 m)



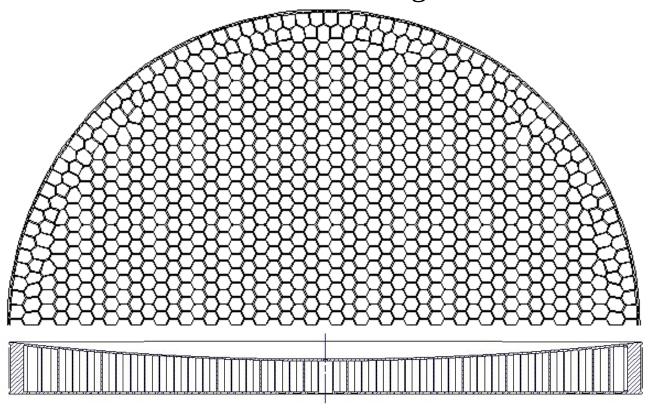


#### Honeycomb mirrors

- Developed by Roger Angel and colleagues at UA
- Concept:
  - Extend Palomar technology to 8 m with more extreme lightweighting
  - Maintain stiffness of traditional mirrors, reducing dependence on active control
  - Achieve very short thermal time constant with thin glass sections, active ventilation
- Technology
  - One-piece spin-casting of honeycomb structure with 80% lightweighting
  - Polishing and measuring very fast mirrors (short focal length, f/1 f/1.25)
- Used for MMT, 2 Magellan telescopes, LBT
- To be used for LSST, GMT 25 m



#### LBT mirror design



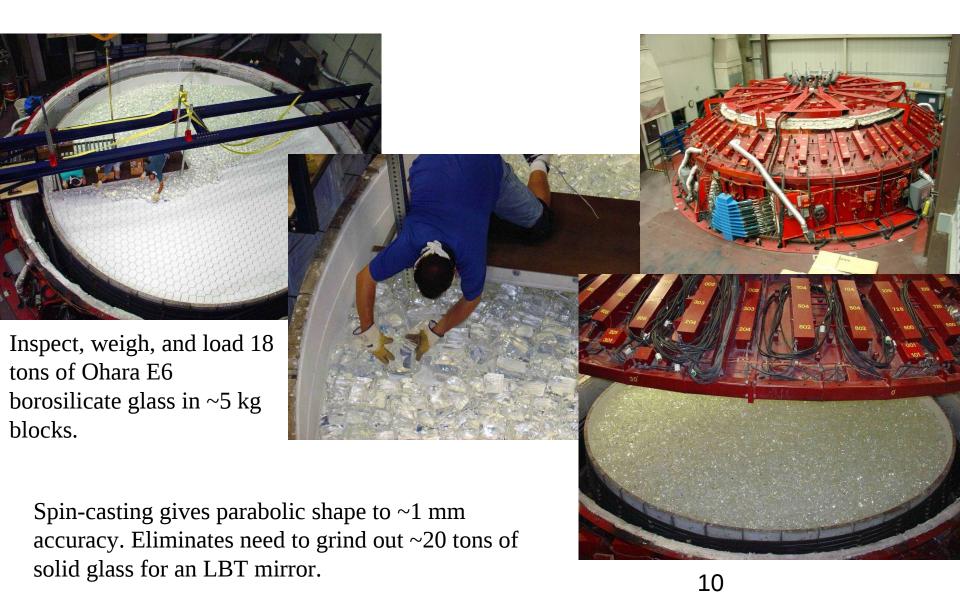
- 1. Borosilicate glass has the lowest expansion coef (3 ppm/K) among materials that can be cast into complex form.
- 2. Facesheet thickness = 28 mm to make  $\tau$  < 1 hr
- 3. Hex cavity size = 192 mm to limit gravity sag of unsupported facesheet to 7 nm
- 4. Rib thickness = 12 mm contributes little mass while maintaining safety.
- 5. Overall thickness 890 mm to give desired stiffness against wind.
- (1) (4) are common to all SOML mirrors; (5) is typical.

## Casting process for GMT mirror: mold assembly



Machine and install 1681 ceramic fiber boxes in silicon carbide tub.

## Loading of glass



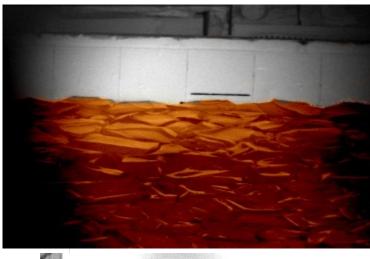
## Glass melting

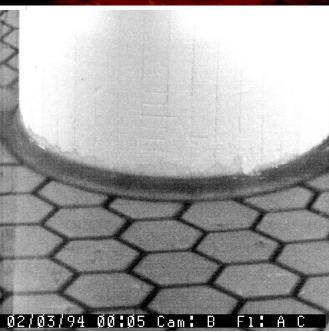
Ø2/Ø2/94 15:Ø6 Cam: B



UV cameras mounted in the furnace lid monitor the casting.

Heat to 1160°C, spin at 4.9 rpm, hold 4 hours to allow glass to fill mold. Cool rapidly to 900°C then slowly for 3 months, 2.4°C/day through annealing.





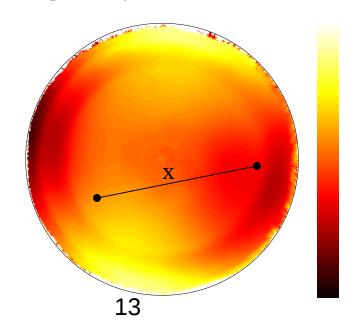
11

First GMT mirror blank

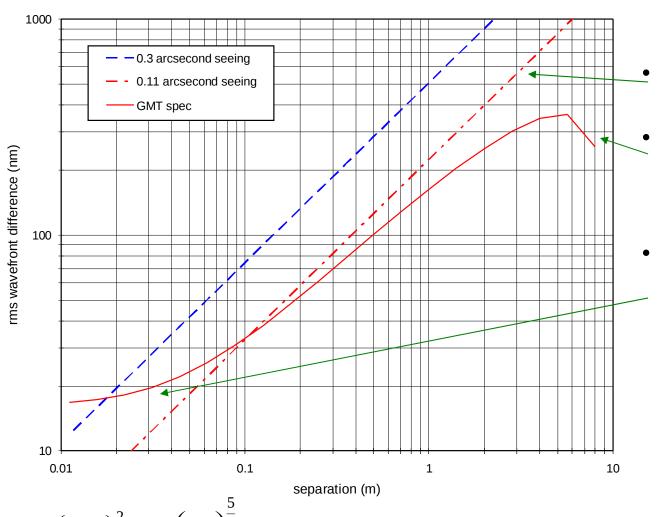


## Manufacturing: Accuracy requirements

- Telescope optics must be more accurate than best wavefront the atmosphere will deliver, at all spatial scales.
  - Without adaptive optics, telescope optics must not significantly degrade images delivered by the atmosphere.
  - With AO, most of DM stroke should be reserved to correct the atmosphere, not the telescope optics.
- "Seeing" is degradation of images due to index variations and turbulence in atmosphere.
  - Typically 0.5 1.0 arcsecond at an excellent site, exceptionally 0.3 arcsecond.
- Atmosphere induces large WF errors on large spatial scales, small errors on small scales.
- Spectrum of WF errors is described by structure function = mean square difference in WF between points in pupil, as a function of their separation *x*.



#### Structure function specification



Start with 0.11 arcsecond seeing.

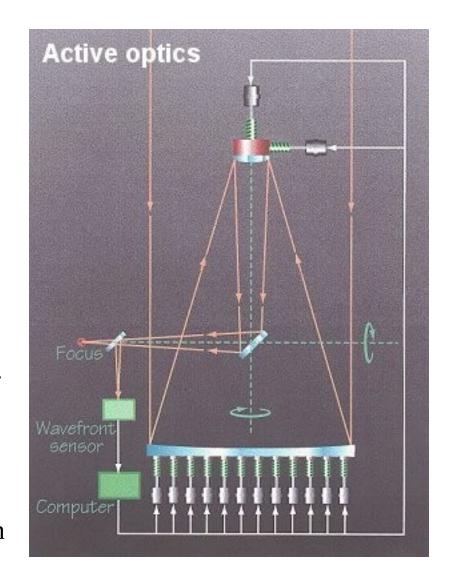
Eliminate tilt across full pupil: tighten on large scales.

Add allowance for 2% scattering loss due to 16 nm rms WFE on small scales.

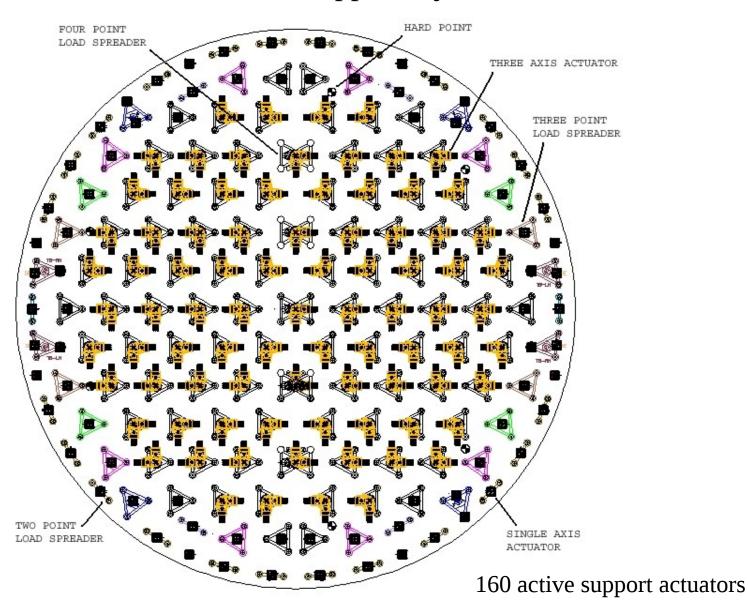
$$\delta^{2}(x) = \left(\frac{\lambda}{2\pi}\right)^{2} 6.88 \left(\frac{x}{r_{0}}\right)^{\frac{5}{3}} \qquad \theta = 0.98 \frac{\lambda}{r_{0}}$$

#### Impact of active optics on requirements

- Active optics is active control of alignment (primary and secondary) and shape of primary, based on WF measurements in telescope.
  - Necessary because no 8 m mirror is rigid
  - Built into all modern telescopes
- Active optics is slow (> 1 minute) and corrects the large-scale errors that are least stable.
- Implication for manufacturing:
  - No need to completely eliminate all loworder shape errors, because they will be controlled with active optics at telescope.
  - In fact, no point in it.
- Manufacturing requirement is to control all aberrations within easy range of active-optics correction in telescope.
- When mirror surface error is measured in lab, simulate active-optics correction of low-order components.

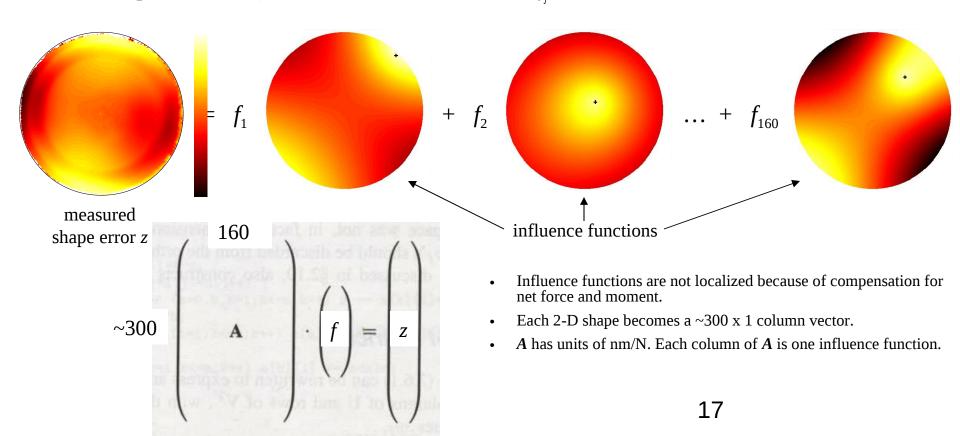


## GMT mirror support layout



#### Active optics as a model-fitting problem

- Calculate or measure the effect on the mirror shape of a unit force on each actuator.
  - → 160 influence functions
- Measure current shape error.
- Find linear combination of influence functions that would match current shape error.
- *Data* are measured surface displacements  $z_i$ . *Model* is sum of influence functions. Model *parameters* (to be determined) are forces  $f_i$ .



## Solving $\mathbf{A}f = \mathbf{z}$

- See Numerical Recipes for insightful (but not easy) description of options.
- Generally have more displacements than forces (more data than unknown model parameters).
- No exact solution: want the approximate solution that minimizes sum of squares of residual errors.
- Find it any number of ways, e. g. Matlab "\" operator, *if you know there are no redundant equations*.
  - 2 or more influence functions that are very similar counts as redundancy.
- If there are, solution will blow up because similar influence functions will be combined with large forces so as to nearly cancel.
- In our case there is redundancy because forces are not independent. They satisfy
  - sum of forces = weight of mirror
  - net moment about x = 0
  - net moment about y = 0
- Could fix that by removing 3 influence functions.
- But generally you also want to limit the forces: remove patterns of forces that contribute little to reducing residual error but use lots of force.
- Take care of both issues, and be much better aware of what's going on physically, by solving with singular-value decomposition....

### Singular-value decomposition

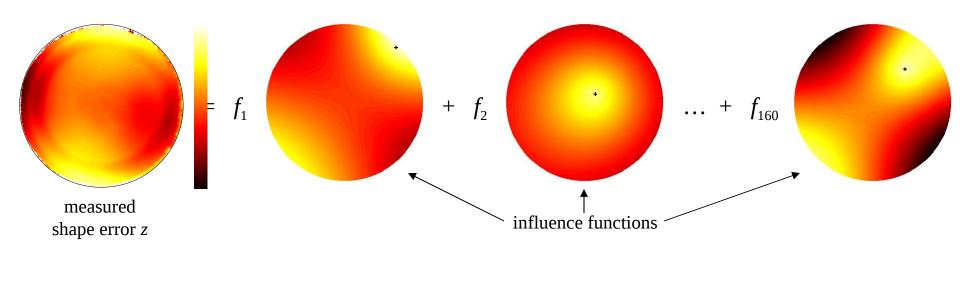
• From *Recipes*:

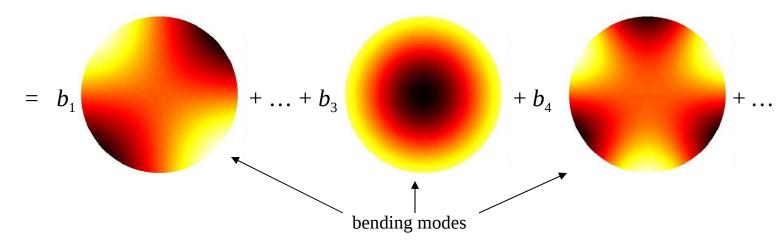
$$\begin{pmatrix} \mathbf{A} & \\ \\ \end{pmatrix} = \begin{pmatrix} \mathbf{U} & \\ \\ \end{pmatrix} \cdot \begin{pmatrix} w_1 & \\ & w_2 & \\ & & \cdots & \\ & & w_N \end{pmatrix} \cdot \begin{pmatrix} \mathbf{V}^T & \\ \\ \end{pmatrix}$$

$$(2.6.1)$$

- Interpret these factors in physical terms:
- *U* has same dimensions as *A*. Columns of *U* are displacement vectors that form an orthonormal basis for all displacements that can be achieved with your 160 actuators.
  - Each column is called a *bending mode*.
- V is 160 x 160. Columns of V are force sets that form an orthonormal basis and match up with columns of U: column j of V produces column j of U.
- W is a 160 x 160 diagonal matrix whose elements  $w_i$  give magnitudes of displacement.
  - If  $f = c_j V_j$  then  $z = c_j w_j U_j$
  - Think of  $w_i$  as the flexibility of bending mode j; it contains all the scaling information.
- SVD is unique apart from re-ordering of columns of U and V, and corresponding  $w_i$ .
- Standard order has  $w_i$  decreasing from most flexible to stiffest.

## Resolve measured shape error into bending modes





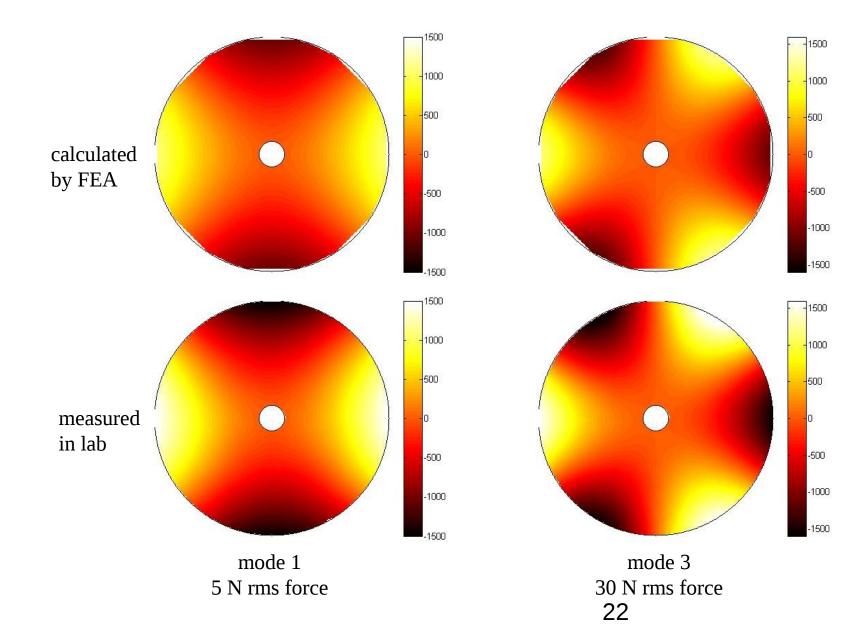
(What does mode 2 look like?)

#### Solution by SVD

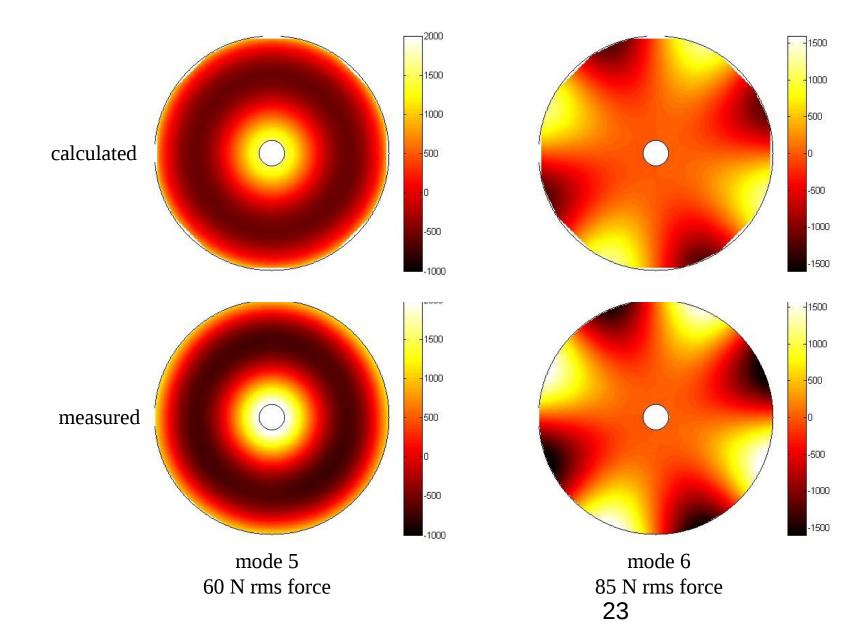
$$\begin{pmatrix} f \end{pmatrix} = \begin{pmatrix} & \mathbf{V} & \end{pmatrix} \cdot \begin{pmatrix} \operatorname{diag}(1/w_j) \end{pmatrix} \cdot \begin{pmatrix} & \mathbf{U}^T & \end{pmatrix} \cdot \begin{pmatrix} \mathbf{z} \end{pmatrix}$$

- From right to left:
  - 1. Take scalar product of measured surface error and each bending mode:
    - Resulting vector tells how much of each bending mode there is (mode coefs  $b_i$  from prev slide).
  - 1. Multiply each mode coefficient by stiffness  $1/w_i$ .
    - Resulting vector tells how much of each force mode.
  - 1. Convert from force modes to actuator forces.
- If there is redundancy, some  $w_i = 0$ .
  - Corresponding columns of *V* are the force vectors that cause no displacement: the *nullspace* of *A*.
  - Can add an infinite amount without changing mirror shape.
  - Eliminate them by setting those  $1/w_i$  to zero.
- Do same for any  $w_j$  small enough to give unreasonable forces.
- Go further: Eliminate all the modes that don't affect the shape enough to justify their large forces.

## Measured bending modes for LBT primary mirror



## Measured bending modes for LBT primary mirror



### Comments on model fitting

- You can solve any model fitting problem in the same way.
  - Measure or calculate the influence of each parameter on the data.
  - Think of it as an influence function, or a sensitivity, or a derivative.
  - E. g. fitting functions to data
    - Influence functions are your functions evaluated at the data points.
    - Solution is the coefficients of the functions.
    - Trivial with, e. g., Matlab "\" operator.
- Be aware of redundancies in model.
  - Use SVD if there are any.
- For SVD, units can matter.
  - SVD minimizes the "length" of the solution vector.
  - If model parameters are of different kinds (e. g., primary mirror support forces and secondary mirror displacements), scale them so a unit change in each is equally "painful".
- Avoid huge range of numbers by normalizing data.
- Think of the problem in physical terms, not just as a system of equations.